# **Concept Map Manual**

### Purpose:

This manual contains the instructions for using the Concept Map software. We will show the features by solving three problems: (i) rosette and maximum, shear strain calculation, (ii) rosette and maximum, shear stress calculation, and (iii) stress-strain calculation for a thin-walled, cylindrical pressure vessel.

## <u>Start-Up</u>

Two concept maps are included in this package: (i) thin-walled, cylindrical pressure vessels, and (ii) stress-strain calculations. We can choose which concept to use as follows:

- $\checkmark$  Choose open from the "File" menu
- ✓ Click radio-button to select map.

Open	Options			
	Accept	] [	Cancel	
	Please s	elect a file to ope	n.	
	C cyl_press_vess.mmf			
	C stress_strain.mmf			
	○ stress_strain.mmf			/

Features of Start-up Window:

- 1. From the "File" drop down menu you can open concept maps and close the program.
- 2. In the 'Utilities' drop down menu, you will find a calculator
- 3. In the 'Mode' drop down menu, you will find and 'Angle' selector which allows you to choose either degrees or radians as your unit of angle measurement.

Three example problems are used to demonstrate the use of the Concept Map tool.

- 1. Maximum shear strain
- 2. Maximum shear stress
- 3. Cylindrical Pressure Vessel

# Calculate Maximum Shear Strain:

A strain rosette, composed of three electrical resistance strain gages making angles of 0, 60, and 120 degrees with the x-axis was mounted on the free surface of a material for which Poisson's ratio is 1/3. Under load, the following strains were measured:

 $\varepsilon_a = +1000\mu$   $\varepsilon_b = +750\mu$   $\varepsilon_c = -650\mu$ 

Determine the principal strains and maximum shear strain.

#### <u>Main Screen</u>

The main screen of the stress-strain concept map is shown in Figure-1.



Figure 1: Stress-Strain Concept Map

The 'spine' of the map includes three nodes: Principal Strains, Hooke's Law, and Principal Stresses. These three nodes represent the three fundamental concepts in the area of stress-strain calculation. The 'right limbs' deal with the plane strain situations, and the "left limbs" deal with the plane stress situations. In addition, there are two more limbs that deal rosettes and elastic properties. This tool will aid you in identifying the most efficient path to take in order to calculate your desired quantities. The tool also identifies ill-posed problems with insufficient information.

#### <u>More Info:</u>

The information contained in each node can be viewed by right-clicking on a node, and clicking "More Info". If we did this for the "Rosette" node, the information contained in the "Rosette" node will be displayed as follows.



### Problem Setup:

✓ Click the 'Display Variables' button to get the following window:



✓ Click the 'View Variables' button

The 'Basic Variables' option is set as the default as seen from the label "Basic" in the drop-down menu at top right.

 $\checkmark$  Check the boxes for the known variables on the left side of the window under the heading 'KNOWN Variables' heading.

Displaying all variables in map	)			
Accept Changes (Enter)	Discard Changes (Esc)	Add Other Variables Basic		
KNOWN va Check all the variab	r <b>iables</b> oles you know.	DESIRED variables Check the one variable you need.		
🔽 Angle-a-theta_a		C Angle-a - theta_a		
🔽 Angle-b - theta_b		C Angle-b - theta_b		
Angle-c-theta_c		C Angle-c-theta_c		
🔽 Poisson's Ratio - nu		C Poisson's Ratio - nu		
🗌 🗖 Shear Modulus - G		🛇 Shear Modulus - G		
🗖 Shear strain - gxy		🛇 Shear strain - gxy		
🗌 🗖 Shear stress - txy		C Shear stress - txy		
🔽 Strain-a - ea		O Strain-a - ea		
🔽 Strain-b - eb		C Strain-b - eb		
🔽 Strain-c-ec		C Strain-c - ec		
🗖 x-strain - ex		C x-strain - ex		
🗖 x-stress - sx		⊂ x-stress - sx		
📃 🗖 Young's Modulus - E		C Young's Modulus - E		
🗖 y-strain - ey		Cy-strain - ey		
🗖 y-stress - sy		⊂ y-stress - sy		
🗖 z-strain - ez		🔿 z-strain - ez		
🗖 z-stress - sz		C z-stress - sz		

The rosette is mounted on the free surface of a material; therefore, the present problem is a planestress problem.

- ✓ Select the 'Control' option in pull-down menu at top right.
  ✓ Click the 'Add Other Variables' button.

D	isplaying all variables in map
	Accept Changes (Enter) Discard Changes (Esc) Add Other Variables Basic
	KNOWN variables Check all the variables you know.
	Plane Strain Control - C_strain
	Plane Stress Control - C_stress

✓ Select 'Plain Strain Control-C\_stress'.

Finally we must select the desired variable.

- ✓ Select 'Derived' from the drop down menu in the upper right
- ✓ Click the 'Add Other Variables' button once.



- ✓ Select "Maximum shear strain gmax" as the desired variable.
- ✓ Click the 'Accept Changes (Enter)' button.

🖳 GenericSelect	
What would you	like to do?
Allow computer to solve the map.	I want to solve the map.

At this point, we will allow the computer to solve the problem for us.

✓ Click 'Allow computer to solve the map' button.

After the application processes the possible solutions, we will be presented with the following window:

postprocessing			
	Post Proc	cessing	
Solution Sorting	Solution Display		
Solution Path	n Cost/Diffic	culty	Effectiveness (%)
Path 1	150	0	•
Use `	This Solution	Clear Probl	em Solutions

✓ Click the 'Use This Solution' button.



The highlighted nodes represent the concepts required for solving the problem.

✓ You can see the information contained in the nodes in the solution path by right-clicking on any node and by selecting "More Info".

On the right hand side of the screen, we see the steps involved in reaching the solution. We will notice that "Step 1" appears three times. When a solution step appears more than once, it means that the step requires solution of simultaneous equations.

✓ Double-click anywhere on the text-block "Step-1: …"

Simultaneous Equations			
			^
Node Rosette			
ex-x-strain	=	0	
🗖 theta_a-Angle-a	=	0	
🗖 ey-y-strain	=	0	
🗖 gxy - Shear strain	=	0	
🗖 ea - Strain-a	=	0	
<u>Node Rosette</u>			
🗖 ex-x-strain	=	0	
🗖 theta_b - Angle-b	=	0	
🗖 ey-y-strain	=	0	≣
🗖 gxy - Shear strain	=	0	
🗖 eb - Strain-b	=	0	
Node Rosette			
🗖 ex-x-strain	=	0	
☐ theta_c - Angle-c	=	0	
🗖 ey-y-strain	=	0	
🗖 gxy - Shear strain	=	0	
🗖 ec - Strain-c	=	0	
		1	
Solve Resolve Clear Solution	F	inished	
Solution:			11 ~

The boxes with the yellow shading represent known variables where the software expects numeric values for the corresponding variables. We cannot enter exponents in these boxes, so  $1000\mu$  has to be entered as 0.001.

- ✓ Enter numeric values for ea, eb, ec, theta\_a, theta\_b, theta\_c.
- ✓ Check ex (Solved variable x-strain)
- ✓ Check ey (Solved variable y-strain)
- ✓ Check gxy (Solved variable shear strain)
- ✓ Click "Solve"

When the solution is complete, the scroll bar on the right will shrink. We can now scroll down to see the calculated values.

Simultane	ous Equations			
•	ex-x-strain	=	0	<u>^</u>
	theta_a-Angle-a	=	0	
☑	ey-y-strain	=	0	
<b>V</b>	gxy - Shear strain	=	0	
Γ	ea - Strain-a	=	.001	
Noc	le Rosette			
<b>V</b>	ex-x-strain	=	0	
Γ	theta_b - Angle-b	=	60	
<b>V</b>	ey-y-strain	=	0	
<b>V</b>	gxy - Shear strain	=	0	
Γ	eb - Strain-b	=	.00075	
Noc	le Rosette			
<b>V</b>	ex-x-strain	=	0	
Γ	theta_c-Angle-c	=	120	≡
<b>V</b>	ey-y-strain	=	0	
$\checkmark$	gxy - Shear strain	=	0	
Γ	ec-Strain-c	=	00065	
So	ve Resolve Clear Solution	F	inished	
Solut	ion:			
ex =	0.00099999999999			
ou =				
ey =				
gxy =	0.00161658075373			//~

After we obtain the solutions to this step, we have several options: (i) we can click 'Resolve' to determine another solution (some problems, such as quadratic equations have multiple solutions). (ii) We can clear the solution, enter new values for the known variables and "Solve" again. (iii) Click 'Finished' tab and move to the next step.

✓ Click the 'Finished' button.

When a step has been completed, its corresponding text on the right side of the screen will turn blue.

✓ Double-click anywhere on the text-box for Step-2

- ✓ Uncheck "Solve Equation-1" (we are not solving for ep1)
   ✓ Uncheck "Solve Equation-2" (we are not solving for ep2)
   ✓ Check "Solve Equation-3" (we are solving for Principal shear strain)
   ✓ Check gp (Solved variable Principal shear strain)
- ✓ Click "Solve"

Solve Equation			(.
Solve equation 2?			^
🗖 ex-x-strain	=	0.0009999	
🗖 ey-y-strain	=	-0.0002666	
🗖 gxy - Shear strain	=	0.0016165	
🗖 ep2 - Principal strain-2	=	0	
Equation 3			
Solve equation 3?			
🗖 ex - x-strain	=	0.0009999	
🗖 ey-y-strain	=	-0.0002666	_
🗖 gxy - Shear strain	=	0.0016165	
🔽 gp - Principal shear strain	=	0	
Solve Resolve Clear Solution		Finished	
Solution:			
			=
			///
gp = 0.00205372312117			/~

- ✓ Click "Finished"
- ✓ Double-click anywhere on the text-box for Step-3

- ✓ Uncheck "Solve Equation-2" (we are not solving for ep1)
  ✓ Check "Solve Equation-1" (we are solving for ep2)
  ✓ Uncheck "Solve Equation-3" (we are not solving for gp)
  ✓ Check "ep2" (Solved variable principal strain-2)
  ✓ Click "Solve"

Solve Equation			
Solve equation 1?			~
🗖 ex - x-strain	=	0.0009999	
🗖 ey - y-strain	=	-0.0002666	
🗖 gxy - Shear strain	=	0.0016165	
🗖 ep1 - Principal strain-1	=	0	
Equation 2			
Solve equation 2?			
🗖 ex-x-strain	=	0.0009999	
🗖 ey - y-strain	=	-0.0002666	
🗖 gxy - Shear strain	=	0.0016165	
🔽 ep2 - Principal strain-2	=	0	
Equation 3			
Solve equation 3?			
🗖 ex-x-strain	=	0.0009999	
🗖 ey - y-strain	=	-0.0002666	
🗖 gxy - Shear strain	=	0.0016165	=
🔲 gp - Principal shear strain	=	0.0020537	
Solve Resolve Clear Solution		Finished	
Solution:			///
			lli
ep2 = -0.0006601949105			//~

- ✓ Click "Finished"
- ✓ Double-click anywhere on the text-box for Step-4

- ✓ Check "Solve Equation-1" (we are solving for ep1)
  ✓ Uncheck "Solve Equation-2" (we are not solving for ep2)
  ✓ Uncheck "Solve Equation-3" (we are not solving for gp)
  ✓ Check "ep1" (Solved variable principal strain-1)
  ✓ Click "Solve"

Solve Equation		
🗖 ey - y-strain	=	-0.0002668
🗖 gxy - Shear strain	=	0.0016165
🔽 ep1 - Principal strain-1	=	0
Equation 2		
Solve equation 2?		
🗖 ex-x-strain	=	0.0009999
🗖 ey - y-strain	=	-0.000266€
🗖 gxy - Shear strain	=	0.0016165
🔲 ep2 - Principal strain-2	=	-0.000660*
Equation 3		
Solve equation 3?		
🗖 ex-x-strain	=	0.0009999
🗖 ey - y-strain	=	-0.0002668
🗖 gxy - Shear strain	=	0.0016165
🔲 gp - Principal shear strain	=	0.0020537
Solve Resolve Clear Solution	L	Finished //
		li
ep1 = 0.00139352821058		// ~
<		> .;;

- ✓ Click the "Finished" button
- ✓ Double Click anywhere on the "Step 5" text

- ✓ Enter Poisson's ratio
  ✓ Check "z-strain"
  ✓ Click "Solve"

Solve Equation		
Plane Stress z-strain		
pl_stress_ez.jpg		
Select one variable per group to solve for:	Give	values to known variables
Equation 1		
Solve equation 1?		
🗖 nu - Poisson's Ratio	=	0.3333
🗖 ex-x-strain	=	0.0009999
🗖 ey-y-strain	=	-0.0002666
C_stress - Plane Stress Control	=	Turned On
💌 ez - z-strain	=	0
Solve Resolve Clear Solution Solution: ez = -0.0003666116527		Finished

- ✓ Click "Finished"
  ✓ Double Click on the "Step 6" text

- ✓ Uncheck Solve Equations1 and Solve Equation 3
  ✓ Check Solve equation 2
  ✓ Check "Intermediate 2"
  ✓ Click "Solve"

Solve Equation		
E ep1 - Principal strain-1	=	0.00139(
🗖 ez - z-strain	=	0
🗖 C_stress - Plane Stress Control	=	Turned C
🗖 pp - Intermediate-1	=	0
Equation 2		
Solve equation 2?		
ep2 - Principal strain-2	=	-0.00066
E ez-z-strain	=	
C_stress - Plane Stress Control	=	
I♥ qq - Intermediate-2	=	lo.
Equation 3		
<u> </u>		
Solve equation 3?		
🗖 gp - Principal shear strain	=	0.00205(
🗖 pp - Intermediate-1	=	0
🔽 qq - Intermediate-2	=	0
C_stress - Plane Stress Control	=	Turned C ≣
🗖 gmax - Maximum shear strain	=	0
Solve Resolve Clear Solution	_	Finished
Solution:		
77 - 0.0000001010		
чч - 0.0006601949		
	_	

✓ Hit "Finished"
✓ Double click on "Step 7" text

- ✓ Uncheck Solve Equation 2 and Solve Equation 3
  ✓ Check Solve equation 1
  ✓ Check "Intermediate 1"
  ✓ Hit "Solve"

Solve Equation			
Select one variable per group to solve for:		Give values to known variables	~
Equation 1 Solve equation 1? ep1 - Principal strain-1 ez - z-strain C_stress - Plane Stress Control pp - Intermediate-1	= = =	0.0013935 0 1 uned On	
Equation 2 Solve equation 2? ep2 - Principal strain-2 ez - z-strain C_stress - Plane Stress Control qq - Intermediate-2	= = =	-0.0006601 0 10med 015 0.0006601	III
Equation 3 Solve equation 3? gp - Principal shear strain pp - Intermediate-1 qq - Intermediate-2 C_stress - Plane Stress Control gmax - Maximum shear strain	= = =	0.0020537 0 0.0006601 Turned Cin- 0	
Solve Resolve Clear Solution		Finished	//~

- ✓ Hit "Finished"
  ✓ Double click the "Step 8" text

- ✓ Uncheck Solve Equation 1 and Solve Equation 2
  ✓ Check Solve Equation 3
  ✓ Check "Maximum Shear Strain"
  ✓ Hit "Solve"

Solve Equation		
		^
Solve equation 1?		
🗖 ep1 - Principal strain-1	=	0.00139
🗖 ez - z-strain	=	0
🗖 C_stress - Plane Stress Control	=	Turned C
🗖 pp - Intermediate-1	=	0.00139
Equation 2		
□ Solve equation 2?		
ep2 - Principal strain-2	=	-0.00066
ez - z-strain	=	0
C_stress - Plane Stress Control	=	Turned C
🗖 qq - Intermediate-2	=	0.000660
<b>F</b> (1)		
Equation 3		
Solve equation 32		3
an - Principal shear strain	_	0.00205
pr - Intermediate-1	_	0.001391
pp Internediate-7	_	1220100.0
C stress - Plane Stress Control	_	Turned f
amax - Maximum shear strain	_	
, ghat maanan shear stain	-	10
Solve Resolve Clear Solution		Finished
Solution:	_	
<		<b>&gt;</b> .::

#### Maximum Shear Stress Concept Map:

A strain rosette makes angles of 0, 45 and 90 with the x-axis. At a point on the free surface of steel (E = 30,000 ksi and v = 0.30) machine part, the strain rosette used to obtain the following normal strain data:

 $\varepsilon a = 650\mu$   $\varepsilon b = 475\mu$   $\varepsilon c = -250\mu$ 

Determine the principal stresses and the maximum shearing stress at the point.

# Problem Setup:

✓ Click the 'Display Variables' button to get the following window:



✓ Click the 'View Variables' button

- ✓ Select the given variables: theta\_a, theta\_b, theta-c, Poisson's ratio, strain-a, strain-b, strain-c and Young's Modulus
   ✓ Select "Control" from the drop down menu
   ✓ Hit the "Add Other Variables" button

Displaying all variables in map	
Accept Changes (Enter) Discard Changes (Esc)	Add Other Variables Basic
KNOWN variables Check all the variables you know.	DESIRED variables Check the one variable you need.
🔽 Angle-a - theta_a	C Angle-a - theta_a
Angle-b - theta_b	C Angle-b - theta_b
Angle-c-theta_c	C Angle-c - theta_c
🔽 Poisson's Ratio - nu	🔿 Poisson's Ratio - nu
🗐 Shear Modulus - G	Shear Modulus - G
🗖 Shear strain - gxy	Shear strain - gxy
🗖 Shear stress - txy	C Shear stress - txy
🔽 Strain-a - ea	O Strain-a ea
🔽 Strain-b - eb	🔿 Strain-b - eb
☑ Strain-c-ec	O Strain-c - ec
🗆 x-strain - ex	C x-strain - ex
□ x-stress - sx	C x-stress - sx
🔽 Young's Modulus - E	🔿 Young's Modulus - E
🗆 y-strain - ey	C y-strain - ey
□ y-stress - sy	⊂ y-stress - sy
🗖 z-strain - ez	🔿 z-strain - ez
🗖 z-stress - sz	C z-stress - sz

Displaying all variables in map	1			
Accept Changes (Enter)	Discard Changes (Esc)	Add Other Variables	Basic 💌	[
KNOWN va Check all the variab	<b>riables</b> les you know.			
📃 🔲 Plane Strain Control - (	C_strain			
🔽 Plane Stress Control -	C_stress			
				1

- ✓ Select "Plain Strain Control-C\_stress"
  ✓ Select "Derived" from the drop down menu
  ✓ Hit the "Add Other Variables" button

Displaying all variables in map	
Accept Changes (Enter) Discard Changes (Esc)	Add Other Variables Basic
KNOWN variables Check all the variables you know.	DESIRED variables Check the one variable you need.
🗖 Maximum shear strain - gmax	C Maximum shear strain - gmax
🗖 Maximum shear stress - tmax	• Maximum shear stress - tmax
🗖 Principal shear strain - gp	Principal shear strain - gp
Principal shear stress - tp	O Principal shear stress - tp
Principal strain-1 - ep1	O Principal strain-1 - ep1
Principal strain-2 - ep2	Principal strain-2 - ep2
Principal stress -1 - sp1     O Principal stress -1 - sp1	
Principal stress-2 - sp2     Principal stress-2 - sp2	
	li l

- ✓ Select 'Maximum shear stress-tmax'
- ✓ Double click on the text for "Step 1"
- ✓ Enter the values for Poisson's Ratio and Young's Modulus
- ✓ Hit "Solve" button



- ✓ Hit the "Finished" button
- ✓ Double click on the text for Step2

- ✓ Check the "z-stress" box
- ✓ Hit "Solve"



- ✓ Hit the "Finished" button
- ✓ Double click on the text for Step 3

- ✓ Enter the "Angle" and "Strain" values
  ✓ Check the box for "x-strain"
  ✓ Check the box for "y-strain"
  ✓ Check the box for "shear strain"

- ✓ Hit "Solve"

Simultaneous Equations			
🔽 gxy - Shear strain	=	0	^
🗖 ea - Strain-a	=	.00065	
Node Rosette			
🔽 ex-x-strain	=	0	
🗖 theta_b - Angle-b	=	45	
🔽 ey-y-strain	=	0	
🔽 gxy-Shear strain	=	0	
🗖 eb - Strain-b	=	.000475	
Node Rosette			
🔽 ex - x-strain	=	0	-
🗖 theta_c - Angle-c	=	90	
🔽 ey-y-strain	=	0	-
🔽 gxy - Shear strain	=	0	-
🗖 ec - Strain-c	=	00025	∃
Solve Resolve Clear Solu	ution	Finished	
Solution:			
ex = 0.00064999999999			
ey = -0.0002499999999			
gxy = 0.00054999999999			~
<		>	1

- ✓ Hit the "Finished" button
  ✓ Double click on the text for Step 4

- ✓ Uncheck Equations 1, 2 and 3
  ✓ Check the box for "Shear stress" in Equation 4
  ✓ Hit "Solve"

Solve Equ	uation				
Г	l gxy - She	ear strain		=	0.00054
3	Solve	Resolve	Clear Solution		Finished
Solutio	)n:				
					≡
txy =	6346.153				
<					

- ✓ Hit the "Finished" button
  ✓ Double click on the text for Step 5

- ✓ Check the boxes for x-stress and y-stress
  ✓ Hit "Solve" button

Simultaneous Equations		
Simultaneous Equations Soluti	on	
Node Hooke's Law		
☑ sx-x-stress	=	0
🗖 nu - Poisson's Ratio	=	<mark>0.3</mark>
🔽 sy-y-stress	=	0
🗖 sz-z-stress	=	0
🗖 E - Young's Modulus	=	30000000
🗖 ex-x-strain	=	0.0006499
Node Hooke's Law		
	=	0
nu - Poisson's Ratio	=	0.3
☑ sx-x-stress	=	0
🗖 sz-z-stress	=	0
🗖 E - Young's Modulus	=	30000000
🗖 ey-y-strain	=	-0.0002499
Solve Resolve Clear Solution		Finished
Solution:		
<sup>sx</sup> = 18956.0439560439		
sy = -1813.1868131868		
		li.

- ✓ Hit the "Finished" button
  ✓ Double click on the text for Step 6

- ✓ Uncheck Equations 1 and 2, leaving Equation 3 checked
  ✓ Select "Principal shear stress"
  ✓ Hit "Solve"

Plane Stress Max Shear Strain		
Solve Equation	_	
☐ sp2 - Principal stress-2	=	0
Equation 3		
Solve equation 3?		
🗖 sx-x-stress	=	18956.04
🗖 sy-y-stress	=	-1813.18
🗖 txy - Shear stress	=	6346.150
🔽 tp - Principal shear stress	=	0
Solve Resolve Clear Solution Solution:		Finished
		E
tp = 12170.2035908932		

- ✓ Hit the "Finished" button
  ✓ Double click on the text for Step 7

- ✓ Uncheck Equations 1 and 3
  ✓ Select "Principal stress-2"
  ✓ Hit "Solve"

					_
Solve Equation				1.	
🗆 🗆 sp1 - Pri	ncipal stress -1		=	0	^
Equation 2					
🔽 <u>Solve eq</u> u	uation 2?				
🗖 sx - x-str	ess		=	18956.04	
			_	,1912.19	
_ sy-y-su	600		-	1013.10	
🗌 🗆 txy - She	ear stress		=	6346.15	
💌 sp2 - Pri	incipal stress-2		=	0	
E-mation 2					
Equation 3					
📃 🗖 <u>Solve eq</u> u	uation 3?				
□ sx - x-str	ess		=	18956.04	
				1010.10	
L sy - y-str	ress		=	-1813.18	
🗌 🗆 txy - She	ear stress		=	6346.15	
🗖 tp - Princ	cipal shear stres	s	=	12170.20	
Solve	Hesolve	Clear Solution		Finished	
Solution:					≣
an2 - Lanas	7000000				
spz =  -3598.77	70908932			11	~
<					

- ✓ Hit the "Finished" button
  ✓ Double click on the text for Step 8

- ✓ Uncheck Equations 2 and 3
  ✓ Select "Principal shear stress-1"
  ✓ Hit "Solve" button

Solve Equation		
		<u>^</u>
Solve equation 2?		
🗖 sx-x-stress	=	18956.04
🗖 sy-y-stress	=	-1813.18
🗖 txy - Shear stress	=	6346.150
🗖 sp2 - Principal stress-2	=	-3598.77
Equation 3		
Solve equation 3?		
🗖 sx-x-stress	=	18956.04
🗖 sy - y-stress	=	-1813.18
🗖 txy - Shear stress	=	6346.15
🗖 tp - Principal shear stress	=	12170.20
Solve Resolve Clear Solution Solution:		Finished
		111
		li
sp1 = 20741.6300908933		

- ✓ Hit the "Finished" button
  ✓ Double click on the text for Step 9

- ✓ Uncheck Equations 1, 2 and 4
  ✓ Select "z-strain" in Equation 3
  ✓ Hit "Solve"

Solve Equation		
	-	<b>^</b>
l✓ ez - z-strain	=	<u>lo</u>
Equation 4		
Solve equation 4?		0040454
Totxy - Shear stress	=	6346.15
G - Shear Modulus	=	1153846
🗖 gxy - Shear strain	=	0.00054
Solve Resolve Clear Solution		Finished
Solution:		
		_
		=
ez =  -0.00017142853		//~
<		> .;;

- ✓ Hit the "Finished" button
  ✓ Double click on the text for Step 10

- ✓ Uncheck Equations 1 and 3
  ✓ Select "Intermediate-2" in Equation 2
  ✓ Hit "Solve"

olve Equation		
Equation 1		<u> </u>
Solve equation 1?		
sp1 - Principal stress -1	=	20741.6
C stress - Plane Stress Control	=	Turned F
	_	0
<ul> <li>bb - interinedicte-i</li> </ul>	-	10
Equation 2		
Solve equation 2?		
□ sp2 - Principal stress-2	=	-3598.77
C stress - Plane Stress Control	=	Turned C
✓ gg - Intermediate-2	_	0
	_	1.
Equation 3		
<u> </u>		
Solve equation 3?		
□ tp - Principal shear stress	=	12170.20
🗖 pp - Intermediate-1	=	0 =
🔽 aa - Intermediate-2	=	0
C stress - Plane Stress Control	=	Turned C
	_	0
	-	
Solve Besolve Clear Solution		Finished
Solution:	_	
Solution.		
44 = [1799.3885		11.~
< III		> 3

- ✓ Hit the "Finished" button
  ✓ Double click on the text for Step 11

- ✓ Uncheck Equations 2 and 3
  ✓ Select "Intermediate-1" in Equation 2
  ✓ Hit "Solve"

Solve Equation		
		^
Solve equation 1?		
🗖 sp1 - Principal stress -1	=	20741.6
🗖 C_stress - Plane Stress Control	=	Turned C
🔽 pp - Intermediate-1	=	0
Equation 2		
Solve equation 2?		
🗖 sp2 - Principal stress-2	=	-3598.77
🗖 C_stress - Plane Stress Control	=	Turned C
🗖 qq - Intermediate-2	=	1799.388
Equation 3		
□ Solve equation 3?		
🗖 tp - Principal shear stress	=	12170.20
🔽 pp - Intermediate-1	=	0
🗖 qq - Intermediate-2	=	1799.388
🗖 C_stress - Plane Stress Control	=	Turned C
🗖 tmax - Maximum shear stress	=	0
Solve Resolve Clear Solution		Finished
Solution:		
PP =  10370.815		<i>I</i> .
<		>

- ✓ Hit the "Finished" button
  ✓ Double click on the text for Step 12

- ✓ Uncheck Equations 1 and 2
  ✓ Select "Maximum shear stress"
  ✓ Hit "Solve"

Solve Equation		
Equation 2		~
□ Solve equation 2?		
sp2 - Principal stress-2	=	-3598.77
C_stress - Plane Stress Control	=	Turned C
🗖 qq - Intermediate-2	=	1799.38
Equation 3		
Solve equation 3?		
🗖 tp - Principal shear stress	=	12170.20
🗖 pp - Intermediate-1	=	10370.8
🗖 qq - Intermediate-2	=	1799.38
C_stress - Plane Stress Control	=	Turned C
🔽 tmax - Maximum shear stress	=	0
Solve Resolve Clear Solution Solution:		Finished
		li
tmax = 12170.2		//~ //~

## **Cylindrical Pressure Vessel:**

A cylindrical pressure vessel with an inside diameter of 1.50 m is constructed by wrapping a 15mm thick steel plate into a spiral and butt-welding the mating edges of the plate. The buttwelded seams form an angle of 30 degrees with a transverse plane through the cylinder. Determine the normal stress perpendicular to the weld and the shearing stress parallel to the weld when the internal pressure in the vessel is 1500 kPa.

#### Problem Setup:

✓ Click the 'Display Variables' button to get the following window:

Displaying all variables in map			
Accept Changes (Enter) Discard Changes (Esc)	Add Other Variables Basic		
<b>KNOWN variables</b> Check all the variables you know.	DESIRED variables Check the one variable you need.		
🗖 Change in radius - dR	Change in radius - dR		
Inclined plane normal angle - theta	C Inclined plane normal angle - theta		
🗖 Poisson's ratio - nu	C Poisson's ratio - nu		
I Pressure - p	O Pressure - p		
🔽 Radius - R	C Radius - R		
□ Rosette angle with axis - theta_r	C Rosette angle with axis - theta_r		
🗖 Rosette strain - er	C Rosette strain - er		
✓ Skin thickness - t	C Skin thickness - t		
✓ Young's modulus - E	C Young's modulus - E		

- ✓ Select the given variables: pressure, radius, thickness, Young's modulus, and theta
- ✓ Select "Derived" from the pull down menu
- ✓ Hit the "Accept Other Variables" button
- ✓ Select "Inclined plain stress –sn"
- ✓ Hit the "Accept Changes [Enter]" button
- $\checkmark$  Allow the computer to find a solution
- ✓ Hit the "Use this Solution" button
- ✓ Double click "Step 1" text

- ✓ Enter the given values for pressure, radius, and skin thickness
  ✓ Select "axial stress"
  ✓ Hit "Solve"

Solve Equation	
Axial Stress	
Select one variable per group to solve for:	Give values to known ∨ariables
Equation 1	
Solve equation 1?	
🗖 p - Pressure	= 1500
🗖 R - Radius	= <mark>.75</mark>
🗖 t - Skin thickness	= <mark>.015</mark>
🔽 sa - Axial stress	= 0
Solve Resolve Clear Solution Solution: sa = 37500	Finished

- ✓ Hit the "Finished" button
  ✓ Double click on the text for Step 2

- ✓ Enter "theta" in the yellow box
  ✓ Uncheck Equation 1
  ✓ Check Inclined plane shear
  ✓ Hit "Solve"

Solve Equation	
Inclined Plane Stresses	
norm_tang.jpg	
Select one variable per group to solve for:	Give values to known variables
Equation 1	
<ul> <li>Solve equation 1?</li> <li>sa - Axial stress</li> <li>theta - Inclined plane normal angle</li> <li>sn - Inclined plane stress</li> </ul>	= 37500 = <mark>30</mark> = 0
Equation 2	
<ul> <li>Solve equation 2?</li> <li>sa - Axial stress</li> <li>theta - Inclined plane normal angle</li> <li>tnt - Inclined plane shear</li> </ul>	= 37500 = 30 = 0
Solve Resolve Clear Solution Solution: tnt = 16237.9763209582	Finished

- ✓ Hit the "Finished" button
  ✓ Double click on the text for Step 3

- ✓ Uncheck Equation 2
  ✓ Select "Inclined plane stress"
  ✓ Hit "Solve"

Our problem is solved.

Solve Equation	
Inclined Plane Stresses	
norm_tang.jpg	
Select one variable per group to solve for:	Give values to known variables
Equation 1	
<ul> <li>Solve equation 1?</li> <li>sa - Axial stress</li> <li>theta - Inclined plane normal angle</li> <li>sn - Inclined plane stress</li> </ul>	= 37500 = <mark>30</mark> = 0
Equation 2	
<ul> <li>Solve equation 2?</li> <li>sa - Axial stress</li> <li>theta - Inclined plane normal angle</li> <li>tht - Inclined plane shear</li> </ul>	= 37500 = <mark>30</mark> = 16237.976
Solve Resolve Clear Solution Solution: sn = 46875	Finished